

## Research Article

# Central configurations of the circular restricted 4-body problem with three equal primaries in the collinear central configuration of the-3 body problem 

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#### Abstract

In this paper we classify the central configurations of the circular restricted 4-body problem with three primaries with equal masses at the collinear configuration of the 3-body problem and an infinitisimal mass


## Introduction and results

The well-known Newtonian n-body problem concerns with the motion of $n$ mass points with positive mass $m_{i}$ moving under their mutual attraction in $\mathrm{R}^{\mathrm{d}}$ in accordance with Newton's law of gravitation.

The equations of the motion of the $n$-body problem are

$$
\ddot{r}_{i}=-\sum_{j=1, j \neq i}^{n} \frac{m_{j}\left(r_{i}-r_{j}\right)}{r_{i j}^{3}}, \quad 1 \leq i \leq n
$$

where we have taken the unit of time in such a way that the Newtonian gravitational constant be one, and $r_{\text {ieR }}{ }^{d}(i=1 \ldots, n)$ denotes the position vector of the $i$-body, $r_{i j}=\left|r_{i}-r_{j}\right|$ is the Euclidean distance between the $i$-body and the $j$-body.

The solutions of the 2-body problem (also called the Kepler problem) has been completely solved, but the solutions for the $n$-body for $n>2$, is still an open problem.

For the Newtonian $n$-body problem the simplest possible motions are such that the configuration formed by the $n$-bodies is constant up to rotations and scaling, such motions are called the homographic solutions of the $n$-body problem, and are the unique known explicit solutions of the n-body problem when $n>2$. Only some special configurations of particles are allowed in the homographic solutions of the n-body problem, called by Wintner [1] central configurations. Also, central configurations are of utmost importance when studying bifurcations of the hypersurfaces of constant energy and angular momentum, for more details see Meyer [2] and Smale [3]. These last years some central configurations have been used for different missions of the spacecrafts in the solar system, see for instance $[4,5]$.

More precisely, let

$$
M=m_{1}+\cdots+m_{n}, \quad c=\frac{m_{1} r_{1}+\cdots+m_{n} r_{n}}{M},
$$

be the total mass and the center of masses of the $n$ bodies, respectively.

